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Multi-objective optimization of nitinol stent design

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ABSTRACT

Nitinol stents continuously experience loadings due to pulsatile pressure, thus a given stent design should possess an adequate fatigue strength and, at the same time, it should guarantee a sufficient vessel scaffolding. The present study proposes an optimization framework aiming at increasing the fatigue life reducing the maximum strut strain along the structure through a local modification of the strut profile.

The adopted computational framework relies on nonlinear structural finite element analysis combined with a Multi Objective Genetic Algorithm, based on Kriging response surfaces. In particular, such an approach is used to investigate the design optimization of planar stent cell.

The results of the strut profile optimization confirm the key role of a tapered strut design to enhance the stent fatigue strength, suggesting that it is possible to achieve a marked improvement of both the fatigue safety factor and the scaffolding capability simultaneously. The present study underlines the value of advanced engineering tools to optimize the design of medical devices.

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1. Introduction

Nowadays, self-expanding nitinol stents are widely used as part of percutaneous minimally-invasive techniques aimed at treating occluded vessels. Unfortunately, several mechanical failures of such a class of devices have been observed [1]; this drawback often results in loss of scaffolding capabilities of the stent, thrombus formation, and restenosis [2,3]. In particular, partial or total stent fractures have been found in aortic [4], renal [5], and pulmonary [6] implants, as well as in lower limb arteries, i.e., superficial femoral artery (SFA) and popliteal artery [7–10]. Therefore, the long-term structural integrity of a given stent model should be one of the principal design parameter to be taken into account.

Given such considerations, it is necessary to optimize stent design performing a thorough engineering analysis, able to assess the relation between the stent geometry and its structural performance. Such an optimization should increase the fatigue strength without penalizing other biomechanical design requirements, such as the vessel scaffolding.

Despite several studies already addressed the analysis of nitinol stents [11–13] and the literature provides some example of stent design optimization analysis [14–19], to the best of our knowledge,

* Corresponding author. E-mail address: gianluca.alaimo01@universitadipavia.it (G. Alaimo). only two studies deal with fatigue strength enhancement of nitinol stents [20,21]. Abad et al. [20] proposed a planar lattice for the realization of a stent with smooth cell shapes in order to reduce peaks of strain induced by abrupt changes in the stent geometry. The study proposes a single-objective optimization process to minimize the curvature of the elementary unit defining the stent design, i.e., stent cell. Azaouzi et al. [21] presented a single-objective optimization approach based on Kriging response surfaces; such an approach has been used to improve the fatigue strength of the stent by minimizing the strut volume without decreasing the stiffness of the stent. The algorithm considers the strut geometry (width, length, and thickness) as the design variables to be optimized. Both studies combine *single-objective* optimization methods with structural finite element analysis (FEA).

Moreover, a further literature analysis suggests that: (i) it is possible to enhance fatigue strength acting on the stent cell design but such an improvement has its counterpart in a loss of stiffness [22,23]; (ii) a tapered strut profile enhances the stent fatigue strength, being thus an ideal starting point for the stent design optimization [24,25].

Relying on the previous observations, in the present study we propose a *multi-objective* optimization procedure acting on both stent cell geometry and strut shape to enhance the fatigue strength of a nitinol stent and its scaffolding capabilities. In particular, the optimization framework accounts for non-linear structural

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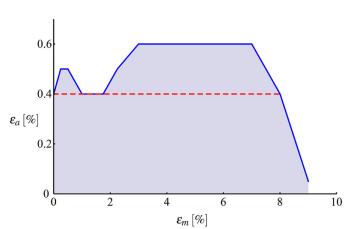


Fig. 1. Constant fatigue life diagram [23,28]: shaded area represents specimens conditions that survived 10^7 cycles while the dashed line represents the value of ε_{a} , i.e., 0.4%, that we adopt in the present paper as conservative threshold.

FEA combined with a Multi Objective Genetic Algorithm (MOGA) [26] based on Kriging response surfaces.

2. Materials and methods

2.1. Fatigue strength analysis

For the purposes of our study we adopt a strain-based approach, which is the most suitable method to deal with fatigue of nitinol stent, as suggested by Robertson et al. [27]. In particular, in case of time-varying cyclic loads, it is possible to define the mean and the alternating value of the first principal strain, respectively ε_m and ε_a , as:

$$\varepsilon_m = \frac{\varepsilon_{\max} + \varepsilon_{\min}}{2} \tag{1a}$$

$$\varepsilon_a = \frac{|\varepsilon_{\max} - \varepsilon_{\min}|}{2} \tag{1b}$$

where $\varepsilon_{\rm max}$ and $\varepsilon_{\rm min}$ are respectively the maximum and the minimum principal strain values in a loading cycle.

As demonstrated by Pelton et al. [23,28], who tested to failure planar diamond-shape specimens under various combinations of ε_m and ε_a , the fatigue strength mainly depends on the alternating value of the first principal strain ε_a . As shown by the diagram depicted in Fig. 1, it is possible to define a conservative threshold of 0.4% for the alternating value of the first principal strain ε_a for any value of mean strain ε_m .

It is worth noting that the damage tolerance analysis (DTA) is the alternative method commonly used for the study of fatigue. Such an approach essentially relies on fracture mechanics and Paris–Erdogan law [29]. As concerns nitinol stents, very few studies have been conducted in order to evaluate the fracture strength and other parameters typical of DTA. One reason is that it holds for values of the flaw size larger than a threshold value below which there is no propagation of the fracture: as reported in Robertson and Ritchie [30] this critical flaw size is about 15–50 µm. Medical devices such as stents have geometric dimensions that are comparable with this threshold value, making DTA useless: it is therefore more useful to focus on prevention rather than on control of the growth of flaws.

For these reasons it can be assumed that the DTA is more appropriate when the typical dimensions of the device are large enough to support the flaw growth or when, for example, the production-process is not sufficiently established to ensure the absence of flaws of critical dimensions [27].

2.2. Stent geometry

In order to define the link between the overall size of a typical "v-shaped" stent and the size of the single cell, it is appropriate to refer to the planar projection of the stent obtained from a virtual unrolling, as illustrated in Fig. 2. In this way, the whole stent design can be thought as a repetition of N cells along the circumference (y axis) and M cells in the axial direction (x axis). For the optimization procedure, we adopt such a simplified planar model. Accordingly, it is possible to define the following geometrical relation:

$$l_c = \frac{\pi D}{N} \tag{2}$$

where D is the outer diameter of the stent and l_c is the length of the unrolled cell in the circumferential direction. Similarly, we have:

$$l_z = \frac{L}{M}$$
(3)

where L and l_z are the lengths of the whole stent and of the cell in the axial direction, respectively.

A change, ΔD , of the stent diameter *D* leads to a variation of the cell height, Δl_c , (see Figs. 2 and 3) that, according to Eq. (2), reads:

$$\Delta l_c = \frac{\pi \, \Delta D}{N} = 2\delta \tag{4}$$

being δ the displacement, along *y*-direction, experienced by the single strut and due to a given variation of the stent diameter ΔD . Each cell is made up of three basic elements: strut, link, and crown. Two struts and the crown that connects them constitute the *v*-shape portion of the planar stent cell [31] as shown in Fig. 2. Fatigue strength and scaffolding do not depend only on the cell geometry but also on strut dimensions (width *w*, length *l*, and thickness *t*) and on its shape (constant cross-sectional profile versus variable one). Thus, we restrict our attention in relating fatigue strength and scaffolding capabilities to such geometrical quantities: lower will be the alternating value of the first principal strain, higher would be the fatigue strength, while higher will be in-plane cell stiffness (ratio *F*/ δ as shown in Fig. 3), higher would be stent scaffolding capability.

To this end, let us consider a v-shape portion of the planar stent cell as shown in Fig.3, subjected to a total displacement $\delta_{\text{tot}} = 2\delta$ in y-direction resulting from the application of a load *F*. We consider the strut as a cantilever beam having rectangular cross-section: from simple beam mechanics, under the assumptions of small strain approximation and linear elastic constitutive behavior [22,23], the maximum elongation in the strut is experienced at the outer curvature. The corresponding first principal strain is:

$$\varepsilon = Z \frac{w \,\delta_{\rm tot}}{l^2} \tag{5}$$

having the value of Z = 3/2 for rectangular cross-section. Eq. (5) shows that the principal strain ε is proportional to the strut width *w* and inversely proportional to the strut length *l*.

We also consider the opening radial force that a nitinol stent applies to the vessel wall after the deployment, namely the chronic outward force (COF) which, in case of "v-shaped" stents, can be evaluated considering the geometry and the mechanics of the single "v-shape" portion [22,23]. Stent and vessel interact with radial forces, acting along the *z*-axis in Fig 2, through the external surface of the stent and the internal wall of the vessel. However, forces applied on the vessel wall by a nitinol stent originates from its circumferential stiffness [22,23] and, in the same way, vessel recoil is contrasted by stent internal circumferential forces (*y*-axis in Fig 2).

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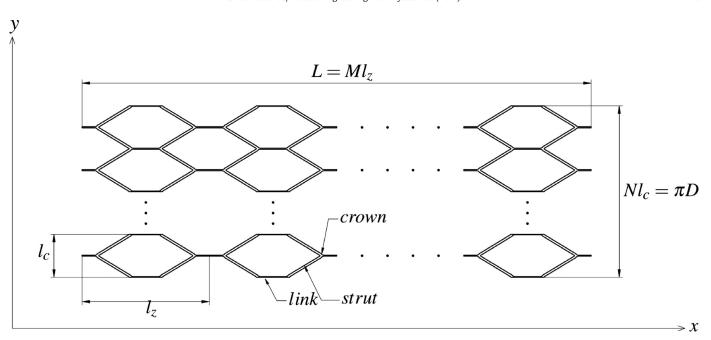
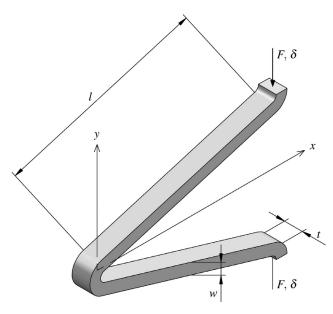


Fig. 2. Planar projection of a typical "v-shaped" stent obtained by a virtual unrolling of the grid: the *x* axis represents the axial direction, the *y* axis represents the circumferential direction while the *z* axis, orthogonal to the *xy* plane, is the radial direction. Global dimensions of the stent and the elements of the cell are shown.



JID: JJBE

Fig. 3. Stent cell "v-shape" portion with dimensions and applied loads. Axes in x and y directions are parallel to those of Fig. 2.

By analogy, pressure acting on a pipe tend to change its diameter: such changes are opposed by the circumferential stiffness of the pipe. According to such considerations, in the present work, we evaluated COF as the force (*F* in Fig 3), in the circumferential direction, resulting from a given "v-shape" displacement δ .

A typical nitinol stent force (per unit cell)-diameter curve is reported in Fig. 4, illustrating the concept of biased stiffness. The stent is crimped at diameter of 3 mm into the delivery system following the loading curve (path A–B). After the release inside the vessel, the stent expands following the unloading path of the curve (path B–A). When the stent reaches the vessel diameter, i.e. 8.3 mm (point C), it continues to push outward against the vessel wall with the equilibrium force COF equal to 0.3 N. On the other hand, vessel recoil is resisted through the force dictated by the reloading curve (path C–D) which is stiffer than the unloading path: the force generated by the stent to resist radial compression (point D) is the ra-

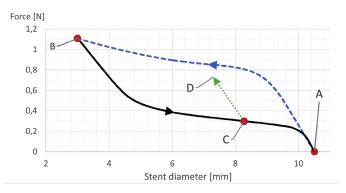


Fig. 4. Force-diameter curve for nitinol stent design *R*. The force is referred to the single cell. Dashed line: loading path A–B, continuous line: unloading path B–A, dotted line: reloading path C–D.

dial resistive force (RRF). If the stent is unloaded, it returns to the unloading path, passing through the point C, but with a hysteresis cycle (not shown in figure). From the analysis of Fig. 4 we can conclude that RRF increases rapidly with diameter changes, while COF is nearly constant at 0.3 N in the neighborhood of point C. Indeed, COF varies from 0.28 N at a diameter of 8.7 mm to 0.31 N at a diameter of 7.9 mm showing a variation less than \pm 5% in this range of diameters. COF is proportional to the cell circumferential stiffness [23] as described by the following equation:

$$\operatorname{COF} \propto \frac{tw^3}{l^3} \tag{6}$$

Eq. (6) shows that COF is proportional to w^3 and inversely proportional to the cube of the strut length *l*. Ideally, we would like to increase COF reducing, at the same time, the maximum principal strain: thus, the considered problem involve two objectives that should be simultaneously optimized. Accordingly, we focus our attention to the class of multi-objective optimization problem.

Remark. The considerations reported in this section are valid only for qualitatively showing the influence of the considered geometrical quantities (t, w, l) on fatigue strength and stiffness of the cell. Indeed, Eqs. (5) and (6) are valid only under the assumption of linear elastic mechanical behavior and small strains approximation.

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Otherwise, the quantity Z in Eq. (5) becomes a function of material and geometry non-linearities. Similar considerations are valid for Eq. (6).

2.3. Multi-objective optimization

In classical optimization problems only one objective function is considered. However, real problems often involve several objectives that should be simultaneously optimized. We consider, without loss of generality, a multi-objective minimization problem whose formulation is:

$$\begin{cases} \min \left\{ f_1(\mathbf{x}), f_2(\mathbf{x}), \dots, f_k(\mathbf{x}) \right\} \\ s.t. \quad \mathbf{x} \in \Sigma \end{cases}$$
(7)

where components $f_i: \Sigma \longrightarrow \mathbb{R}$, i = 1, 2, ..., k are the objective functions and $\Sigma \subset \mathbb{R}^n$ is the feasible design domain. Furthermore, we define the objective space $\Omega \subset \mathbb{R}^k$ as the image of Σ , under the mapping function $\mathbf{f} = [f_1(\mathbf{x}), f_2(\mathbf{x}), ..., f_k(\mathbf{x})]$; the elements $\boldsymbol{\omega} \in \Omega$ are the objective vectors. A design vector $\hat{\mathbf{x}}$ and the corresponding solution $\hat{\boldsymbol{\omega}} = \mathbf{f}(\hat{\mathbf{x}})$ of problem (7) are defined trivial if:

$$\hat{\omega}_j = f_j(\hat{\mathbf{x}}) = \min_{\mathbf{x}\in\Sigma} \left\{ f_j(\mathbf{x}) \right\}, \quad \forall j = 1, 2, \dots, k$$
(8)

according to definition (8), the design vector $\hat{\mathbf{x}}$ is defined trivial if it minimizes simultaneously all the objective functions f_j , $j = 1, 2, \ldots, k$. Problem (7) is defined *nontrivial* if does not exist a trivial solution and, in this case, the objective functions are said to be conflicting. When problem (7) is *nontrivial*, there exists a (possibly infinite) number of optimal solutions that can be identified introducing the definition of dominance.

A design vector $\hat{\mathbf{x}}_1$ dominates another design vector $\hat{\mathbf{x}}_2$ if both the following conditions are true:

$$f_i(\mathbf{\hat{x}_1}) \le f_i(\mathbf{\hat{x}_2}), \quad \forall i = 1, 2, \dots, k$$

$$\tag{9a}$$

$$\exists j \text{ such that } f_i(\mathbf{\hat{x}}_1) < f_i(\mathbf{\hat{x}}_2) \tag{9b}$$

Eq. (9a) states that the design vector $\hat{\mathbf{x}}_1$ is no worse than $\hat{\mathbf{x}}_2$ for all objectives and Eq. (9b) states that design vector $\hat{\mathbf{x}}_1$ is strictly better than $\hat{\mathbf{x}}_2$ in at least one objective.

Relying on the concept of dominance, the Pareto set is defined as the set of non-dominated solutions of problem (7). From a mathematical point of view, every design in the Pareto set is an equally optimal solution of the multi-objective optimization problem. We conclude noting that we would like to increase COF reducing, at the same time, the maximum principal strain, but the analysis of Eqs. (5) and (6) reveals that such objectives are conflicting. Accordingly, we search for optimal solutions within the Pareto set. The selection of one design from the Pareto optimal set can be done choosing the trade-off point, i.e. the Pareto optimal point that is the most appropriated to the design context [32].

2.4. Alternative design

In this section we introduce an alternative design, characterized by the use of a tapered strut, in order to reduce the maximum value of the first principal strain and preserving COF, at the same time. Although it has been observed that the use of tapered profile strut may improve fatigue strength of nitinol stent [25] and such an idea is even covered by patent [24], we note that in the scientific literature, to the best of our knowledge, there are not (i) any specific information about the optimal geometry with respect to the fatigue strength and COF and (ii) available experimental data concerning this type of design.

As an illustrative example we consider the cantilever parabolicprofile, as shown in Fig. 5, whose width w(x) is function of the

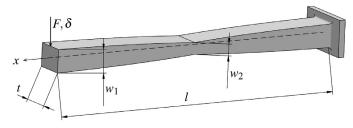


Fig. 5. Cantilever beam with symmetric parabolic profile.

abscissa x, as described by the following relation:

$$w(x) = w_1 \left[\sqrt{1 - \frac{x}{l}} H\left(\frac{l}{2} - x\right) + \sqrt{\frac{x}{l}} H\left(x - \frac{l}{2}\right) \right]$$
(10)

where w_1 is the height of the tips, *l* the beam length and H(x) is the Heaviside unit step function defined as:

$$H(z) = \begin{cases} 1 & \text{if } z > 0\\ 0 & \text{if } z \le 0 \end{cases}$$
(11)

The selected parabolic-profile has been chosen to obtain a constant value of the first principal strain (and stress) along the first half of the beam, i.e. for 0 < x < l/2. Relation (10) shows that $w_1/w_2 = \sqrt{2}$, where $w_2 = w(l/2)$ is the width at the center of the beam.

In order to capture the impact of the design on the strut principal strain, we compare its maximum value for the tapered and for the uniform profiles. The comparison is provided assuming the same length l and stiffness $k = F/\delta$ for both solutions, namely k_c for uniform profile and k_p for the tapered one, being F and δ the force and the displacement at the free end, respectively.

The analysis is performed relying on the methodology proposed by Romano [33]. Most recent works [34,35] highlight that such an approach must be carefully considered because tapered beams could show a very different mechanical behavior with respect to prismatic ones. However, given the qualitative nature of the present example and considering that we are examining beam exhibiting slow cross-section variations ($\left|\frac{dw}{dx}\right| < 0.1 \quad \forall x \in [0, l]$), the adopted approach may be considered sufficiently accurate. Summing up, for the constant-section beam of width *w*, the stiffness is:

$$k_c = \frac{tEw^3}{4l^3} \tag{12}$$

while, for the parabolic profile, the beam stiffness is:

$$k_p = \frac{tEw_1^3}{8l^3}$$
(13)

where *t* and *E* are the thickness and the Young's modulus of the beam, respectively. The same stiffness for both designs can be obtained imposing the equality of Eqs. (13) and (12), yielding $w_1 = \sqrt[3]{2} w \simeq 1.26 w$.

The expression of the principal strain $\varepsilon_c(x)$ for the uniform profile reads:

$$\varepsilon_c(x) = \frac{3w\delta}{2l^3}(l-x) \tag{14}$$

while, for the parabolic one, we obtain the following relation:

$$\varepsilon_{p}(x) = \begin{cases} \frac{3w\delta}{2l^{2}} \frac{1}{\sqrt[3]{4}} & \text{if } 0 \le x \le \frac{l}{2} \\ \frac{3w\delta}{2l^{2}} \frac{1}{\sqrt[3]{4}} \left(\frac{l}{x} - 1\right) & \text{if } \frac{l}{2} < x \le l \end{cases}$$
(15)

The expressions of the normalized principal strain along the nondimensional coordinate x/l are shown in Fig. 6. Both solutions have

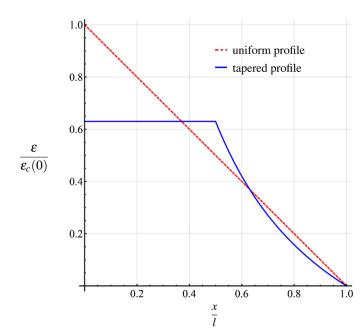


Fig. 6. Graph of the deformation along the *x* axis normalized with respect to the maximum that occurs in the constant-section beam $\varepsilon_c(0)$.

their maximum for x = 0 and their ratio is:

$$\frac{\varepsilon_p(0)}{\varepsilon_c(0)} = \frac{1}{\sqrt[3]{4}} \simeq 0.63 \tag{16}$$

Eq. (16) shows that, keeping the same stiffness, length l, and thickness t, the considered parabolic-profile beam (10) has a maximum value of strain which is substantially lower than the one experienced by the constant-section beam. Additionally, Fig. 6 shows that, in case of constant section, the principal strain ε_c varies linearly along the beam axis while the parabolic profile (10) exhibits a constant distribution of the first principal strain along the first half of the beam and then ramps down to zero with a hyperbolic law. Clearly, when considering profiles different to (10), strain distribution along the beam x-axis will be different but, from a qualitative point of view, well-designed tapered profiles contribute to uniform strain field and exhibit lower values of the maximum first principal strain, compared to prismatic ones. Consequently, we conclude that acting on geometrical quantities such as l, w_1, w_2 , it is possible to define enhanced geometries that may provide longterm fatigue strength when considered in a structural optimization context.

Remark. Parabolic profile (10) only depends on two independent parameters: we choose the width w and the length l of the beam. Accordingly, the corresponding beam stiffness (13) does not depend on w_2 . The ratio w_1/w_2 can be varied only changing the functional relationship (10).

2.5. Finite element model

The considerations reported in Sections 2.1 and 2.4 are used to generate a virtual model of the stent-cell to be used in the structural FEA. Numerical simulations are performed using the commercial software Ansys.

The optimization of the stent cell is obtained in terms of control variables corresponding to the geometrical features of the tapered profile, collected in the vector \mathbf{x} defined as (see Fig. 7):

$$\mathbf{x} = [w_1, l, a, b, c]^T$$

where *l* is the length of the strut, w_1 the width at the ends of the strut, and *a*, *b* and *c* dimensionless variables defined as:

$$a = \frac{w_2}{w_1} \tag{18a}$$

$$b = \frac{r}{w_1} \tag{18b}$$

$$c = \frac{w_k}{w_1} \tag{18c}$$

being w_2 the width at the midpoint of the strut, r the crown outer radius and w_k the width of the link. Nitinol stents are usually lasercut from a standard tube using a crimped design and then expanded and heat-treated to reach the final diameter. In crimped design, the thickness of the strut is usually constant for the convenience of the laser-cutting process and the saving of expensive materials. If the thickness is changed, the diameter of the compact design has to be also changed, thus the original tube will not be compatible. For the mentioned reason, during the optimization process, the thickness t is kept fixed, i.e. $t = 200 \,\mu\text{m}$.

The overall cell dimensions l_c and l_z are assumed constant during the optimization procedure because they depend on the number of cells in the circumferential and the longitudinal directions, namely *N* and *M*, as defined by Eqs. (2) and (3) respectively. Indeed, variations of these dimensions may result in an excessive variation of the cell surface that is an important feature for the stent performance.

Cell geometry is also defined by the angle ϕ and the length of the link l_k (see Fig. 7) that are related to the chosen values of l_c and l_z and to the input variables (Eq. (17)) as:

$$l_c = 2l\cos\phi + 2(2b - 1)w_1\sin\phi \tag{19a}$$

$$l_{z} = 2l_{k} + 2l\sin\phi + 4\left[bw_{1} - w_{1}\left(b - \frac{1}{2}\right)\cos\phi\right]$$
(19b)

that resolved with respect to ϕ and l_k yield:

$$\phi = \arctan\left[\frac{l(l_c w_1(2b-1) + A)}{l^2 l_c + (1-2b)w_1 A}\right]$$
(20a)

$$l_{k} = \frac{l_{z}}{2} - l\sin\phi - 2\left[bw_{1} - w_{1}\left(b - \frac{1}{2}\right)\cos\phi\right]$$
(20b)

where $A = l \left(4l^2 - l_c^2 + 4(1 - 2b)^2 w_1^2\right)^{1/2}$. From Eqs. (20a) and (20b), it can be observed that the parameters *a* and *c* do not affect the overall geometry of the cell, but they only have a local effect, in particular on widths w_2 and w_k , respectively. The strut profile is represented by a spline curve.

Nitinol super-elastic behavior is simulated by the Shape Memory Alloy (SMA) model available in Ansys [36]; the SMA characteristics have been assumed coincident with those reported by Meoli et al. [37] and shown in Fig.8. More in detail, σ_s^{AS} and σ_f^{AS} are the values of stress at the start and at the end of the austenite-martensite (AM) transformation, σ_s^{SA} and σ_f^{SA} are the initial and final stress in the martensite-austenite (MA) transformation, E_a and E_m are the moduli of elasticity of austenite and martensite, ν is the Poisson's ratio and ε_L is the maximum equivalent strain at the end of the AM transformation. The parameter α takes into account the different behavior in tension and compression and it is defined by the following equation:

$$\alpha = \frac{\left(\sigma_s^{AS}\right)_c - \left(\sigma_s^{AS}\right)_t}{\left(\sigma_s^{AS}\right)_c + \left(\sigma_s^{AS}\right)_t} \tag{21}$$

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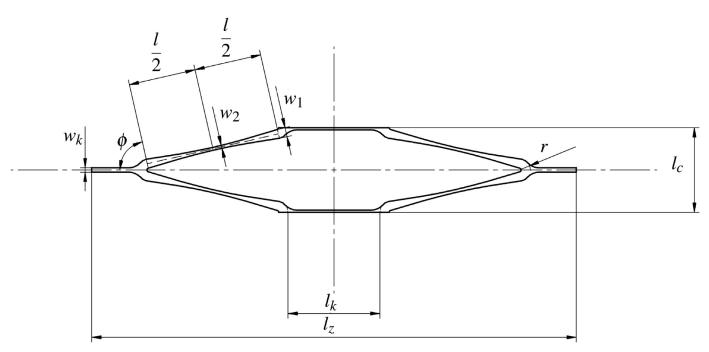


Fig. 7. Parametric model of the stent-cell.

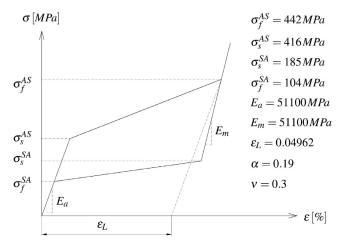


Fig. 8. Curve $\sigma - \varepsilon$ for an ideal super-elastic material.

where $(\sigma_s^{AS})_c$ and $(\sigma_s^{AS})_t$ are respectively the compressive and tensile stress at the start of the AM transformation. The cell model is discretized with the 8-node brick element Solid185 available in Ansys element library, with full integration. A convergence analysis was performed to ensure a suitable mesh refinement. The quantity selected during the convergence analysis was the maximum value of the first principal strain on the whole strut, after the crimping inside the delivery system (end of step 1). The analysis was performed decreasing the average element size in step of 1 µm in the range 20–10 µm and using several geometries of the strut. The best trade-off between solution accuracy and efficiency was obtained using an element size of 12 µm. The adopted mesh involves a number of elements which varies, approximately, between 30000 and 40000 according to the considered design.

Fig. 9a shows the whole cell and its geometrical planes of symmetry α , β and γ . We simulated only the strut imposing symmetry boundary conditions on the surfaces resulting from the intersection of the cell with such planes. In particular (see Fig. 9b), on the surface *F* parallel to the *xy* plane, displacements δ along *z*, due to the changes in diameter *D*, are imposed. On the surface $A \subset \beta$ parallel to the *xy* plane, displacements in the *z* direction are pre-

vented, on the surface $B \subset \gamma$ parallel to the *xz* plane, displacements in the *y* direction are prevented, while on the surface $C \subset \alpha$ parallel to the *yz* plane, displacements in the *x* direction are prevented.

The stent under investigation is characterized by a diameter D = 10.5 mm and a length L = 42 mm; then, considering a number of cells equal to N = 14 and M = 7 in the circumferential and axial directions respectively, by the relations (2) and (3) we obtain the circumferential length l_c and the axial length l_z of the cell as $l_c = 2.4$ mm and $l_z = 7$ mm. The considered loading history (Fig. 10) involves four distinct steps:

- Step 1. Stent crimping: the stent diameter goes from diameter $D_0 = 10.5 \text{ mm}$ to diameter $D_1 = 3 \text{ mm}$. It follows (Eq. (4)) that the displacement in the *z* direction of the surface *F* (Fig. 9b) reads $\delta_1 = 841 \,\mu \text{m}$.
- Step 2. Stent deployment into the vessel (femoral artery): the stent diameter goes from diameter $D_1 = 3 \text{ mm}$ to diameter $D_2 = D_s = 9.6 \text{ mm}$ which coincides with the diameter of the stent at the end of the systolic phase [23]. The resulting displacement, in agreement with the Eq. (4), is equal to $\delta_2 = 101 \,\mu\text{m}$.
- Step 3. Contraction of the stent due to the diastolic blood pressure: the stent diameter goes from $D_2 = 9.6$ mm to diameter $D_3 = D_d = 9.3$ mm coincident with the diameter at the end of the diastolic phase; the displacement at the end of this step is equal to $\delta_3 = 135 \,\mu$ m.
- Step 4. Expansion of the stent: return to systolic diameter $D_4 = D_s = 9.6$ mm, corresponding to the cycle end.

In order to evaluate the change in diameter $\Delta D = D_s - D_d$ in steps 3 and 4, resulting from a pressure variation Δp , we set the arterial compliance to be (after [38]):

$$C = 100 \, \frac{D_s - D_d}{D_d} = 3.26\% \tag{22}$$

corresponding to a variation in the blood pressure $\Delta p = 100 \text{ mmHg}$; similar compliance values are obtained in [39]. According to [38], the corresponding cross sectional arterial stiffness K_A , i.e. the inverse of the compliance per unit of length of

Fig. 9. Finite element model. (a) Planes of symmetry α , β and γ for the unit cell. (b) FE-model with boundary conditions and some details of the mesh.

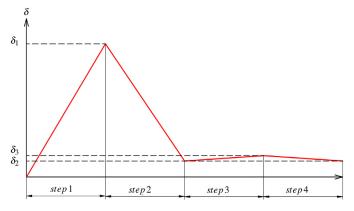


Fig. 10. Displacement history imposed on strut free end. Step 1: stent crimping, step 2: stent deployment, step 3, step 4: stent contraction and expansion, respectively, due to blood pressure variation.

the vessel is:

$$K_A = \frac{dp}{dA} = \left(\frac{1}{l}\frac{dV}{dp}\right)^{-1} = 12.1 \frac{\text{mmHg}}{\text{mm}^2}$$
(23)

where A, l and V are the cross sectional area, the length and the volume of the vessel, respectively.

The quantities of interest are the alternating maximum value of the first principal strain ε_a and the chronic outward force COF. In order to evaluate them, we coded an Ansys Parametric Design Language (APDL) script. More in detail, at the end of steps 3 and 4 for each element of the model the values of the first principal strain are evaluated, and then the mean values ε_m and the alternating value ε_a are calculated: the maximum value of ε_a is stored for each design point **x**. Similarly, a quantity proportional to COF, represented by the force acting on the surface *F* (indicated in Fig. 9b) at the end of step 2 and necessary to keep the displacement δ_2 , is computed and stored.

Remark. As observed in Section 2.2, COF is the opening force that the nitinol stent applies to the vessel wall after the deployment. It can be indifferently evaluated at the end of step 2 or step 4, but not at the end of step 3, because of the biased stiffness of nitinol stents.

2.6. Kriging response surfaces

For completeness, a brief description of Kriging surfaces (KS) [40–42] is given. KS predict the response $z(\mathbf{x})$ of a function $f(\mathbf{x})$ at unobserved design points $\mathbf{x} \in \Sigma \subset \mathbb{R}^n$ relying on the known values $f(\tilde{\mathbf{x}}^i)$ at all the sampled points $\tilde{\mathbf{x}}^i$, i = 1, 2, ..., M.

We define $\tilde{\mathbf{f}}$ as the vector of dimension *M*, that contains the values of the function $f(\mathbf{x})$ at each sampled point, namely $\tilde{\mathbf{f}} = [f(\tilde{\mathbf{x}}^1), f(\tilde{\mathbf{x}}^2), \dots, f(\tilde{\mathbf{x}}^M)]^T$. The general expression of KS is:

$$z(\mathbf{x}) = d(\mathbf{x}) + r(\mathbf{x}) \tag{24}$$

where $d(\mathbf{x})$ is a polynomial function of \mathbf{x} , and $r(\mathbf{x})$ is the realization of a normally distributed Gaussian random process with zero mean, variance σ_r^2 and non-zero covariance. The term $d(\mathbf{x})$ in Eq. (24) is usually termed as "trend" as it globally approximates the unknown function $z(\mathbf{x})$ over the feasible design space Σ , while the term $r(\mathbf{x})$ allows for local deviations, enabling KS to interpolate the *M* sampled points. The term $d(\mathbf{x})$ is defined as:

$$d(\mathbf{x}) = \mathbf{q}^T(\mathbf{x})\,\mathbf{v} \tag{25}$$

where $\mathbf{q}(\mathbf{x}) = [q_1(\mathbf{x}), q_2(\mathbf{x}), \dots, q_m(\mathbf{x})]^T$ is the polynomial basis vector, which dimension *m* depends on the degree of $d(\mathbf{x})$ -2 in the present work- and on the dimension *n* of the design space Σ ; $\mathbf{v} = [v_1, v_2, \dots, v_m]^T$ is a vector of unknown coefficients. The covariance matrix **C** of $r(\mathbf{x})$ is a *M* x *M* matrix, whose elements are given by:

$$C_{ij} = \sigma_r^2 R_{ij} \tag{26}$$

where R_{ij} are the elements of **R** representing the spatial correlation functions between each pair $(\tilde{\mathbf{x}}^i, \tilde{\mathbf{x}}^j)$ of the sampled points. The spatial correlation function used in the present work is a Gaussian correlation function defined as:

$$R_{ij} = R(\tilde{\mathbf{x}}^i, \tilde{\mathbf{x}}^j) = \exp\left(-\sum_{k=1}^n \lambda_k \left|\tilde{x}^i_k - \tilde{x}^j_k\right|^2\right)$$
(27)

where λ_k , k = 1, 2, ..., n, are unknown correlation parameters and the quantities \tilde{x}_k^i and \tilde{x}_k^j are the k^{th} components of the sampled points $\tilde{\mathbf{x}}^i$ and $\tilde{\mathbf{x}}^j$, respectively. Given such assumptions, the vector of unknown coefficients **v** in Eq. (25) is obtained by least square

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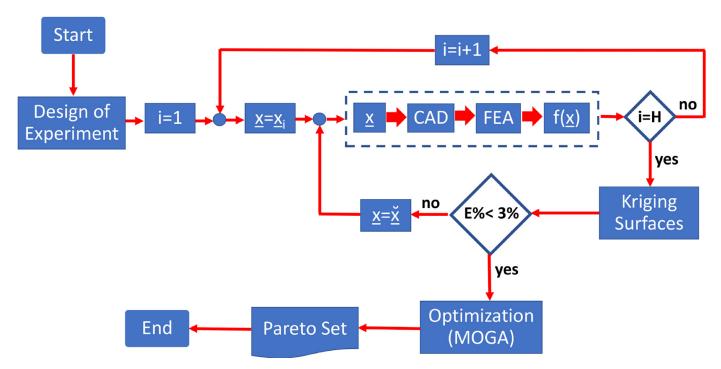


Fig. 11. Flow chart of the optimization process. A single cycle is described by the following steps: generation of the geometry (CAD) obtained by design variable (\mathbf{x}), finite elements analysis (FEA), assessment of the objectives $\mathbf{f}(\mathbf{x})$. Both design of experiment and Kriging surfaces refinement steps generate design points \mathbf{x} and evaluate the corresponding values of the objective function $\mathbf{f}(\mathbf{x})$ by FEA, while the optimization process, performed by MOGA, is based on Kriging surfaces.

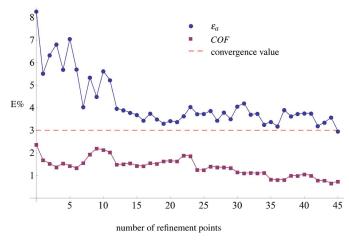


Fig. 12. Graph of the maximum expected relative error E^{∞} versus number of refinement points relative to the Kriging surfaces. It is shown that 45 additional refinement points were needed to reach the selected value of 3% for the maximum predicted relative error. Red dashed line represents the convergence value. (For interpretation of the references to colour in this figure legend, the reader is referred to the web version of this article.)

regression, obtaining:

$$\tilde{\mathbf{v}} = \left(\mathbf{Q}^T \, \mathbf{R}^{-1} \, \mathbf{Q}\right)^{-1} \mathbf{Q}^T \, \mathbf{R}^{-1} \, \tilde{\mathbf{f}}$$
(28)

where \mathbf{Q} is the *M* x *m* matrix defined as:

$$\mathbf{Q} = \begin{bmatrix} \mathbf{q}^{T}(\tilde{\mathbf{x}}^{1}) \\ \mathbf{q}^{T}(\tilde{\mathbf{x}}^{2}) \\ \vdots \\ \mathbf{q}^{T}(\tilde{\mathbf{x}}^{M}) \end{bmatrix}$$
(29)

KS in Eq. (24) is thus obtained as:

$$z(\mathbf{x}) = \mathbf{q}^{T}(\mathbf{x})\,\tilde{\mathbf{v}} + \mathbf{g}^{T}(\mathbf{x})\,\mathbf{R}^{-1}\left(\tilde{\mathbf{f}} - \mathbf{Q}\,\tilde{\mathbf{v}}\right)$$

where g(x) is the correlation vector between the point x and the sampled data points, namely:

$$\mathbf{g}(\mathbf{x}) = [R(\mathbf{x}, \tilde{\mathbf{x}}^1), R(\mathbf{x}, \tilde{\mathbf{x}}^2), \dots, R(\mathbf{x}, \tilde{\mathbf{x}}^M)]^T$$
(31)

Correlation parameters λ_k in Eq. (27) are to be determined before the KS can be computed. They are evaluated maximizing, over λ_k , k = 1, 2, ..., n, the following functional Φ :

$$\Phi(\lambda_1, \lambda_2, \dots, \lambda_n) = M \ln(\sigma^2) + \ln(\det \mathbf{R})$$
(32)

where σ^2 is the estimate of variance σ_r^2 in Eq. (26), that is given by:

$$\sigma^{2} = \frac{\left(\tilde{\mathbf{f}} - \mathbf{Q}\tilde{\mathbf{v}}\right)\mathbf{R}^{-1}\left(\tilde{\mathbf{f}} - \mathbf{Q}\tilde{\mathbf{v}}\right)}{M}$$
(33)

The predicted deviation of the KS from the actual response, for each point $\mathbf{x} \in \Sigma$, is statistically represented by the root mean squared error $E(\mathbf{x})$, that is defined as:

$$E(\mathbf{x}) = \sqrt{\sigma^2} \begin{bmatrix} 1 - \begin{bmatrix} \mathbf{q}^T(\mathbf{x}) & \mathbf{g}^T(\mathbf{x}) \end{bmatrix} \begin{bmatrix} \mathbf{0} & \mathbf{Q}^T \\ \mathbf{Q} & \mathbf{R} \end{bmatrix} \begin{bmatrix} \mathbf{q}(\mathbf{x}) \\ \mathbf{g}(\mathbf{x}) \end{bmatrix}}$$
(34)

Typical steps in defining KS are: the evaluation of correlation parameters λ_k by maximizing non-linear functional (32), determination of the correlation matrix **R** by (27), evaluation of KS (30) through (28) and (31), error estimation (34).

2.7. Optimization

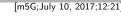
We assume that the domain $\Sigma \subset \mathbb{R}^5$ of the multi-objective optimization is bounded within the following intervals:

$$w_1 \in [80, 200]$$
 (35a)

$$l \in [1500, 2500]$$
 (35b)

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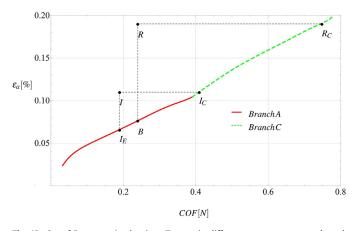


Fig. 13. Set of Pareto optimal-points. Two main different zone are present: branch *A* and branch *C*. Point *R* represents the constant-section strut design, point *B* represents the optimized tapered strut and point *I* represents the same geometry of the design *R* but with tapered profile.

$$a \in [0.35, 1]$$
 (35c)

 $b \in [1.2, 1.5]$ (35d)

$$c \in [0.5, 1.5]$$
 (35e)

where w_1 and l are expressed in μ m and a, b and c are dimensionless. The range of variation of the input variables, has been chosen considering the typical dimensions of the strut composing commercial nitinol stents. In particular, we chose as reference design \mathbf{x}_R the Smart stent by Cordis [43] (see [23,37]), having a constant cross section strut (a = 1) with length l, height w_1 and thickness tequal to 2000 μ m, 120 μ m and 200 μ m, respectively; other parameters b and c are equal to 1.25 and 1 respectively. Accordingly, the reference design \mathbf{x}_R is obtained with our parametric model, employing the following design variables:

$$\mathbf{x}_{R} = [w_{1}, l, a, b, c]^{T} = [120, 2000, 1, 1.25, 1]^{T}$$
(36)

The array of the objective functions is defined as:

$$\mathbf{f}(\mathbf{x}) = \left[\varepsilon_a(\mathbf{x}), \operatorname{COF}(\mathbf{x})\right]^{\prime}$$
(37)

We preliminarily run a FEA to simulate the reference design \mathbf{x}_{R} , obtaining an alternating strain ε_{a} equal to 0.19% with COF = 0.24 N (point *R* in Fig. 13). Assuming a fatigue limit of $\varepsilon_{L} = 0.4\%$ as discussed in Section 2.1, for the reference design the safety factor yields:

$$S_R = \frac{\varepsilon_L}{\varepsilon_a} = \frac{0.4}{0.19} = 2.1 \tag{38}$$

The multi-objective optimization aims at minimizing $\varepsilon_a(\mathbf{x})$ and, at the same time, maximizing COF(\mathbf{x}), with $\mathbf{x} \in \Sigma$. Furthermore, the additional constraint $\varepsilon_a \leq 0.2\%$ is imposed to admit only \mathbf{x} (i.e. designs) corresponding to a value of the fatigue safety factor $S \geq 2$: in this way we consider only solutions having safety factor comparable with or higher than the reference design \mathbf{x}_R . The mathematical formulation of the considered optimization problem reads:

Objectives:
$$\begin{cases}
 \min \{\varepsilon_a(\mathbf{x})\} \\
 max \{COF(\mathbf{x})\}
 \end{cases}$$
(39)Constraints:
$$\begin{cases}
 \varepsilon_a(\mathbf{x}) \leq 0.2 \\
 \mathbf{x} \in \Sigma
 \end{cases}$$
(40)

The goal of the multi-objective optimization is to evaluate the Pareto set: to obtain it we use a MOGA [26,44] based on KS. In the present work we use KS for each component of the objective vector $\mathbf{f}(\mathbf{x})$. KS are preliminarily built during the domain mapping phase (see Fig. 11), which is essentially the selection of a set of points \mathbf{x} in which $\mathbf{f}(\mathbf{x})$ is evaluated by FEA. Such initial set is obtained with the Central Composite Design (CCD) method [44]. According to this approach, the objective vector $\mathbf{f}(\mathbf{x})$ is evaluated for H design points, with $H = 2^m + 2m + n_c = 54$ where m is the number of input variables (m = 5), and $n_c = 12$ is the number of auxiliary points that allow for estimation of second-order effects.

Once the initial mapping of $\mathbf{f}(\mathbf{x})$ is completed (design of experiment step in Fig. 11), a first KS interpolation of the sampled points is performed, for each objective function. The greater is the number of points used to construct the surface, the lower will be its deviation from the exact value. An error estimator of KS is obtained introducing the maximum predicted relative error E_i %, defined as:

$$E_i \approx \frac{100}{f_i^{\max} - f_i^{\min}} \max_{\mathbf{x} \in \Sigma} [E_i(\mathbf{x})] = \frac{100}{f_i^{\max} - f_i^{\min}} E_i(\mathbf{\tilde{x}}_i), \quad i = \varepsilon_a, \text{COF}$$
(41)

where f_i^{max} and f_i^{min} are the maximum and the minimum known values (on design points **x**) of the objective f_i under study, respectively, $E_i(\mathbf{x})$ is the root mean square error defined in Eq. (34) and $\mathbf{\tilde{x}}_i$ is the design point for which the error $E_i(\mathbf{x})$ is maximum. Clearly, the error E_i % is not the same for both the objective functions: we define the error $E\% = \max_i [E_i\%]$ and $\mathbf{\tilde{x}}$ the point in which E% occurs. We opt for a value of the maximum predicted relative error of 3% as best trade-off between computational time needed for solving KS and accuracy. In order to obtain the requested value of E%, for each response surface refinement step, we generate the additional refinement points $\mathbf{\tilde{x}}$, and we use the corresponding $\mathbf{f}(\mathbf{\tilde{x}})$, calculated by FEA, to update KS. This iterative process is concluded when the convergence is reached, as shown in Fig. 12.

As depicted in Fig. 11, once the refinement of KS is obtained, i.e., the response surface phase is completed, the optimization step can be performed.

3. Results

The obtained Pareto front, reported in Fig. 13, shows two branches, namely *A* and *C*, characterized by different trends of input variables; in particular, the part *A* is generated by the set of arrays \mathbf{x}_A such that:

$$\mathbf{x}_{A} = [w_{1}, l, a, b, c]^{T} = [w_{1}, 2500, 0.5 - 0.6, 1.5, 1.5]^{T}$$
(42)

with w_1 varying from 80 µm to 200 µm for increasing values of COF. Relation (42) states that branch *A* is only function of the width of the strut w_1 , that may be increased when more scaffolding capabilities are requested. As concerns the length of the strut *l*, the crown radius *b* (Eq. (18b)) and the width of the link *c* (Eq. (18c)), their optimal values are obtained increasing them as much as possible, i.e. choosing their maximum allowed in the considered range (see Eq. (35)). Moreover, we observe that the optimal value of the input parameter *a*, that represents the ratio between the width at the center and at the end of the strut (see Eq. (18a)), varies among 0.5–0.6; consequently, the optimal solutions are obtained when the strut profile is not constant.

The branch *C* is generated by the the set of arrays \mathbf{x}_{C} such that:

$$\mathbf{x}_{\rm C} = [w_1, l, a, b, c]^{\rm T} = [200, l, 0.6 - 0.64, 1.5, 1.5]^{\rm T}$$
(43)

with the length *l* decreasing from $2500 \,\mu\text{m}$ to about $1700 \,\mu\text{m}$ for increasing values of COF. Relation (43) states that branch *C* is only function of the length of the strut *l*, that may be decreased when

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higher scaffolding is required. Similarly to the case of branch A, optimal designs in branch C are obtained by increasing the width of the strut w_1 , the crown radius b and the width of the link c up to their maximum value. The optimal value of the parameter a varies in the range 0.6–0.64.

Better results should be obtained allowing the optimization for higher value of the input variables l, w_1 , b and c in the design space Σ . However, such extension will involve the change of the length and of the height of the unit cell, that are fundamental parameters in stent design. The parameter a for the Pareto set shows small oscillations in the reported ranges for branches A and C, without a well-defined trend. Accordingly, we considered a nearly constant. It is also important to note that the Pareto optimal set is always represented by tapered strut profiles with a ratio between the width at the center and at the end of the strut ranging in the interval 0.5–0.65.

4. Discussion

In order to highlight the improvements in stent fatigue strength achieved by the proposed optimization analysis, we compare the performance of our resulting stent design with the reference design \mathbf{x}_{R} .

We note that all the points falling within the area BRR_c , depicted in Fig. 13, are represented by design parameters that improve, compared to the reference design, the fatigue strength or the scaffolding, or both simultaneously. Moreover, all the points that are on the portion of the Pareto front delimited by points *B* and R_c , represent, by definition, the solutions for which it is not possible to further improve the fatigue strength without decreasing scaffolding capabilities and vice-versa. Accordingly, such solutions are equivalent and all of them are optimal designs (in Pareto sense) in the domain Σ : the best design trade-off can be selected by restricting the attention to the Pareto set.

As explanatory case study, we show how we maximize fatigue strength, selecting an optimal solution from the Pareto set, under the requirement of leaving COF unchanged with respect to the design \mathbf{x}_{R} . Such optimal solution (Fig. 13) is represented by the point *B* defined as:

$$\mathbf{x}_{B} = [w_{1}, l, a, b, c]^{T} = [163, 2493, 0.57, 1.49, 1.4]^{T}$$
(44)

for which we obtain an alternate strain $\varepsilon_a = 0.079\%$ and, consequently, the fatigue safety factor yields:

$$S_B = \frac{\varepsilon_L}{\varepsilon_q} = \frac{0.4}{0.079} = 5.1$$
 (45)

Comparing Eqs. (38) and (45) we observe that, with the same scaffolding capabilities, introducing the optimized tapered profile (in this case $w_1 = 163 \,\mu\text{m}$, $w_2 = 93 \,\mu\text{m}$, $l = 2493 \,\mu\text{m}$), it is possible to increase the fatigue safety factor under pulsatile loads of approximately 2.4 times.

The effect induced by the introduction of the tapered profile only in the reference design \mathbf{x}_R is investigated considering the design resulting from the following input parameters:

$$\mathbf{x}_{l} = [w_{1}, l, a, b, c]^{l} = [120, 2000, 0.57, 1.25, 1]^{l}$$
(46)

The design \mathbf{x}_{l} differs from the reference design \mathbf{x}_{R} (Eq. (36)) only for the value of the parameter a = 0.57 selected in correspondence with the absolute minimum point for the alternating strain ε_{a} . Therefore, the corresponding geometries are equal, except for the strut profile, which is tapered in the design \mathbf{x}_{l} . For such design (Fig. 13) we find an alternating strain $\varepsilon_{a} = 0.11\%$ and a value of the COF equal to 0.19N. Assuming a fatigue limit of $\varepsilon_{L} = 0.4\%$ the fatigue safety factor reads:

$$S_l = \frac{\varepsilon_L}{\varepsilon_a} = \frac{0.4}{0.11} = 3.6\tag{47}$$

Table 1

Geometrical features and value of the objectives for the typical (\mathbf{x}_R) , optimized (\mathbf{x}_B) and tapered (\mathbf{x}_I) design.

Design	w ₁ (μm)	<i>l</i> (μm)	а	b	с	\mathcal{E}_a (%)	COF (N)
\mathbf{x}_R	120	2000	1	1.25	1	0.19	0.24
\mathbf{x}_B	163	2493	0.57	1.49	1.4	0.079	0.24
\mathbf{x}_{l}	120	2000	0.57	1.25	1	0.11	0.19

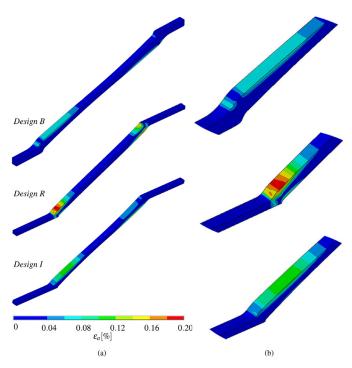


Fig. 14. (a) Contour plot of the alternating strain ε_a for different designs of the cell: design *B* represents the optimized profile, design *R* represents typical *b* (constant section) profile while design *I* is equal to *R* except for the tapered profile. (b) Details of the proposed designs. From top to bottom: design *B*, *R* and *I*.

whereas the ratio between the COF corresponding to design points \mathbf{x}_I and \mathbf{x}_R is 0.19/0.24 = 0.79. Then, we conclude that only considering the tapered profile allows to increase significantly the fatigue strength but, at the same time, it involves a relative loss of scaffolding capabilities, approximately equal to 20%. Furthermore, as observed as regard to the design \mathbf{x}_R , the design \mathbf{x}_I does not belong to the Pareto set and then it is not a optimal solution in the domain Σ : indeed with the same scaffolding of \mathbf{x}_I it can be obtained a design with better fatigue strength (point I_E in Fig. 13) or, with the same fatigue strength, better scaffolding capabilities may be reached (point I_C in Fig. 13). The designs considered in this section are reported in Table 1.

From an analysis of Fig. 14 it is evident that the parts of the cell with higher risk of failure are near to the external part of the crown, as confirmed by other authors [45]. Moreover, the tapered profile allows to redistribute the strain along a wider part of the strut yielding a decrease of its maximum value as discussed in Section 2.4.

In Fig. (15) we show the resulting stent unit cell for both designs \mathbf{x}_R and \mathbf{x}_B . The length and the height of the unit cell are the same for both designs, while the corresponding geometries differ for the strut profile, crown radius and link width.

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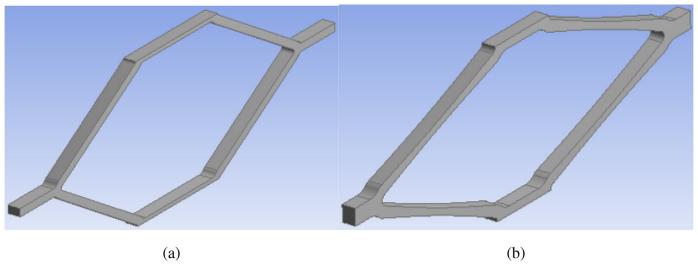


Fig. 15. Stent cell geometries. (a) Typical design R. (b) Optimized design B.

5. Limitations

The major limitation of this study lies in the execution of numerical simulations considering the entire cell in a planar configuration. As already highlighted by other authors [19,46,47], it may be considered sufficiently accurate although (i) the actual complete 3D model has curved shape and (ii) stents interact with vessels, resulting in a mechanical response influenced by contact phenomena. Results much closer to the in vivo conditions may be obtained by optimizing complete models of stents. In this regard, it might be useful to carry out experimental tests aimed at validating the proposed optimized design.

It is worth noting that, in addition to the time needed to implement the optimization process (parametric model, boundary and load conditions, post-processing of results), the considered problem requires non-trivial computation times; then, more realistic study require computational platforms such supercomputing.

6. Conclusions

The present study discusses the use of multi-objective stent design optimization to enhance fatigue life of self-expanding nitinol stent and vessel scaffolding capability. The results obtained through the proposed optimization study are related to a vessel with assigned compliance and blood pressure variation. The methodology introduced is still valid and can be applied also to different pressure cycles and anatomical positions. The study results confirm that the use of tapered strut profile should be a primary key factor to reduce and uniform the strain field along the strut and thus to enhance the fatigue life of the whole stent. The obtained Pareto set allows the designer for the selection of optimized solution, according to the specific design requirements.

As illustrative example we compared a commercial reference design with an optimized design, chosen from the obtained Pareto set, under the requirement of leaving COF unchanged. The proposed approach suggests that the enhancement of stent fatigue life can be achieved combining tapered strut profile with the following changes in the design of the cell:

- an increase of 25% of the strut length;
- an increase of 40% of the strut width at the strut extremities.

Moreover, the results suggest that the width narrowing at the middle of the strut, due to the profile tapering, should be stay among 35–50%.

Under such indications, it is possible to achieve a marked improvement of the fatigue safety factor, i.e., about 2.4 times, compared to the typical design (strut with constant section), without any loss of scaffolding capabilities.

The present study may be used as a starting point for further optimization analyses addressing the design of brand-new peripheral stent models. Further developments can address extension approach to a full 3D case or experimental validation of the achieved results by the performance of fatigue tests for the proposed stent strut design.

Conflict of interest

None.

None.

Funding

Ethical approval

Not required.

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